

UNIVERSITY OF SWAZILAND**FACULTY OF SCIENCE****DEPARTMENT OF PHYSICS****SUPPLEMENTARY EXAMINATION 2005**

TITLE OF THE PAPER: CLASSICAL MECHANICS

COURSE NUMBER : P320

TIME ALLOWED : THREE HOURS

INSTRUCTIONS:ANSWER ANY FOUR OUT OF FIVE QUESTIONS.EACH QUESTION CARRIES 25 MARKS.
MARKS FOR DIFFERENT SECTIONS ARE SHOWN IN THE
RIGHT-HAND MARGIN.

THIS PAPER HAS 4 PAGES, INCLUDING THIS PAGE.

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GIVEN PERMISSION.

Q.1. The Lagrangian for a system can be written as

$$L = a\dot{x}^2 + b\frac{\dot{y}}{x} + c\dot{x}\dot{y} + f y^2 \dot{x}\dot{z} + g\dot{y} - k\sqrt{x^2 + y^2}$$

where a, b, c, f, g and k are constants.

- (i) List the generalized coordinates. [2]
- (ii) Write down the equations of motion. [10]
- (iii) What is the Hamiltonian? [10]
- (iv) What quantities are conserved? [3]

Q.2. (i) A particle of mass m is under the influence of potential [7]

$$V(r) = -\frac{k}{r}. \text{ Show that the total energy } E = \frac{m\dot{r}^2}{2} + \frac{\ell^2}{2mr^2} - \frac{k}{r}.$$

(ii) If the launching velocity of a satellite from the surface of the earth [10]
is $v < v_e$, where v_e is the escape velocity, show that the maximum
height h that can be obtained for vertical launch is given by

$$h = \frac{2R^2 g}{2Rg - v^2}$$

where R is the radius of the earth.

(iii) Show that for a geo-stationary satellite of mass m and at [8]
distance r from the centre of earth is given by

$$r^3 = \frac{GMT^2}{4\pi^2}$$

where G = gravitational interaction constant, M = mass of earth and T is the
period of revolution.

Q.3. (i) Bounded orbits are possible only for attractive potential $V(r) = -\frac{k}{r}$

and orbits are ellipse or circle with two turning points r_1 and r_2 .

Semi-major axis $a = \frac{r_1 + r_2}{2}$ and semi minor axis $b = a\sqrt{1 - \varepsilon^2}$ where

$$\varepsilon = \sqrt{1 + \frac{2E\ell^2}{mk^2}} \text{ where } \ell \text{ is orbital angular momentum.}$$

Show that $\tau^2 = \frac{4\pi^2 m}{k} a^3$ where τ is the period of the orbit and m is the [15]
reduced mass.

(ii) The major axis of the elliptic orbit of a certain comet is 100 astronomical
units.

- (a) What is its period? [3]
- (b) If its distance from the sun is $\frac{1}{2}$ astronomical unit at perihelion, [5]
what is the value of the eccentricity of the orbit?
- (c) What is the comets's speed at perihelion and aphelion? [2]

Note: Assume that mass of sun is very large compared to the mass of comet.

Q.4. (A) Consider free small oscillations in one dimension about a position of stable equilibrium of a particle with mass m and displacement as x .

(i) Show that the potential is given by $\frac{1}{2} k x^2$. [5]

(ii) Derive an expression for the Lagrangian. [2]

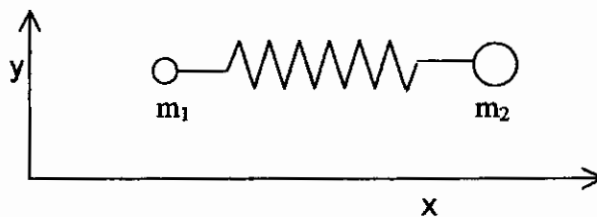
(iii) Derive the equations of motion for this system. [2]

(iv) Show that angular frequency depends only on the property of the mechanical system. [2]

(v) Show that the total energy $E = \frac{1}{2} m \omega^2 a^2$, where $\omega = \sqrt{\frac{k}{m}}$ [3]

and a is the amplitude.

(B) A typical diatomic molecule may be regarded as equivalent to two masses m_1 and m_2 connected by a massless spring of spring constant k and of unstretched length a , vibrating along the line joining the two masses as shown



Taking coordinates of m_1 and m_2 as x_1 and x_2 respectively,

(i) write down the expressions for potential and kinetic energy. [2]

(ii) write down the equation of motion. [3]

(iii) show that the system has two normal angular velocities [8]

$$\omega_1 = 0 \text{ and } \omega_2 = \pm \left(\frac{k(m_1 + m_2)}{m_1 m_2} \right)^{1/2}$$

Q.5. (A) To an observer in the rotating system, rotating with angular velocity ω , a moving particle is under the influence of an effective force

$$\vec{F}_{eff} = \vec{F} - 2m(\vec{\omega} \times \vec{V}_r) - m\vec{\omega} \times (\vec{\omega} \times \vec{r})$$

where \vec{F} = force in the inertial system,

\vec{V}_r = velocity of the particle in the rotating set of axes.

(i) Which is the term in the above expression corresponding to Coriolis force? [1]

(ii) What is the physical interpretation of the last term? [1]

(iii) Earth rotates with $\omega = 7.292 \times 10^{-5} \text{ rad/s}$. Calculate the value of centripetal acceleration at the equator where $r = 6.378 \times 10^6 \text{ m}$.

Explain the effect of this acceleration on a falling body. [3]

- (iv) At the Northern Hemisphere, what is the direction of ω . [5]
 Fixing the z-axis along this direction, diagrammatically illustrate the deflection of a projectile shot along the earth's surface.
 (v) What will be the direction of the projectile if shot along the earth's surface at Southern Hemisphere and at the Equator? [2]

(B) For a force free motion of a symmetrical top, Euler equations are

$$I_1 \dot{\omega}_1 = (I_1 - I_3) \omega_3 \omega_2$$

$$I_2 \dot{\omega}_2 = - (I_1 - I_3) \omega_3 \omega_1$$

$$I_3 \dot{\omega}_3 = 0$$

and $\omega^2 = \omega_1^2 + \omega_2^2 + \omega_3^2$.

- (i) Show that $\omega_1 = A \cos(\Omega t)$ and $\omega_2 = A \sin(\Omega t)$ [7]

Where $\Omega = \frac{(I_3 - I_1)}{I_1} \omega_3$ and A is a constant.

- (ii) Show that magnitude of $\bar{\omega}$ is constant. [2]

- (iii) The earth can be approximated as symmetrical about the polar axis and slightly flattened at the poles so that $I_1 < I_3$.

Numerically, $\frac{I_3 - I_1}{I_1} \approx 0.00327$.

Assuming $\omega_3 \approx \omega$, calculate the period predicted for the precession of the axis of rotation [4]