

**UNIVERSITY OF SWAZILAND**

127

**FACULTY OF SCIENCE**

**DEPARTMENT OF PHYSICS**

**MAIN EXAMINATION 2011/2012**

**TITLE OF PAPER : CLASSICAL MECHANICS**

**COURSE NUMBER : P320**

**TIME ALLOWED : THREE HOURS**

**INSTRUCTIONS : ANSWER ANY FOUR OUT OF FIVE  
QUESTIONS.  
EACH QUESTION CARRIES 25  
MARKS.  
MARKS FOR DIFFERENT SECTIONS  
ARE SHOWN IN THE RIGHT-HAND  
MARGIN.**

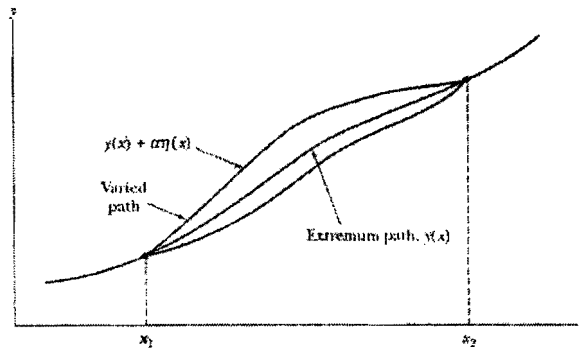
**THIS PAPER HAS NINE PAGES, INCLUDING THIS PAGE.**

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**P320 CLASSICAL MECHANICS**

**Question one**

- (a) Given the following definite integral of  $J(\alpha) = \int_{x_1}^{x_2} f(y(\alpha, x), y'(\alpha, x), y''(\alpha, x); x) dx$ , where the varied integration path is  $y(\alpha, x) = y(x) + \alpha \eta(x)$ ,  $\eta(x_1) = \eta(x_2) = 0$  and  $\left. \frac{d\eta(x)}{dx} \right|_{x=x_1} = \left. \frac{d\eta(x)}{dx} \right|_{x=x_2} = 0$  as shown in the following diagram :



Using the extremum condition for  $J(\alpha)$ , i.e.,  $\left. \frac{\partial J(\alpha)}{\partial \alpha} \right|_{\alpha=0} = 0$ , to deduce that

$f$  along the extremum path, i.e.,  $f(y(x), y'(x), y''(x); x)$ , satisfies the following equation:

$$\frac{\partial f}{\partial y} - \frac{d}{dx} \left( \frac{\partial f}{\partial y'} \right) + \frac{d^2}{dx^2} \left( \frac{\partial f}{\partial y''} \right) = 0 \quad (13 \text{ marks})$$

- (b) For a certain dynamical system the kinetic energy  $T$  and potential energy  $V$  are given by

$$T = \frac{1}{2} (\dot{q}_1^2 + \dot{q}_1 \dot{q}_2 + \dot{q}_2^2)$$

$$V = \frac{3}{2} q_2^2$$

where  $q_1, q_2$  are the generalized coordinates.

- (i) Write down Lagrange's equations of motion. (6 marks)
- (ii) Deduce an expression for  $q_2$  in terms of  $t$ . (6 marks)

**Question two**

- (i) Show that for two dimensional motion of a particle of mass  $m$  with a constant acceleration  $\alpha$  along  $+x$  direction and a zero acceleration along  $y$  direction the Hamiltonian is given by the expression,

$$H = \frac{p_x^2}{2m} + \frac{p_y^2}{2m} - m\alpha x \quad (6 \text{ marks})$$

- (ii) From the definition of the Poisson brackets, i.e.,  $[u, v] \equiv \sum_{\alpha} \left( \frac{\partial u}{\partial q_{\alpha}} \frac{\partial v}{\partial p_{\alpha}} - \frac{\partial u}{\partial p_{\alpha}} \frac{\partial v}{\partial q_{\alpha}} \right)$ , evaluate  $[x, H]$  and  $[p_x, H]$ . (8 marks)

- (iii) For an equation of the type  $\frac{du}{dt} = [u, H]$  the specific solution of  $u(t)$  is given by the following series expansion

$$u(t) = u_0 + [u, H]_0 t + [[u, H], H]_0 \frac{t^2}{2!} + [[[u, H], H], H]_0 \frac{t^3}{3!} + \dots$$

where subscript 0 denotes the initial conditions at  $t = 0$ .

Use the above relation to show that for the given Hamiltonian, the specific solution of  $x(t)$  is given by

$$x(t) = x_0 + \frac{p_{x,0}}{m} t + \frac{\alpha}{2} t^2$$

where  $x_0$  and  $p_{x,0}$  are the initial x-position and x-momentum respectively.

(11 marks)

### Question three

- (a) Given the Lagrangian for the two-body central force system as :

$$L = \frac{1}{2} \mu (\dot{r}^2 + r^2 \dot{\theta}^2) - U(r)$$

where  $\mu$  is the reduced mass of the system and  $(r, \theta)$  are polar coordinates of the motion plane with its origin at the center of mass of the two-body system.

- (i) Write down the Lagrange's equation for  $\theta$  and show that the angular momentum  $l$  is conserved, i.e., deduce that

$$\dot{\theta} = \frac{l}{\mu r^2} \quad \dots\dots (1) \quad \text{where } l \text{ is a constant.} \quad \text{(3 marks)}$$

- (ii) Write down the Lagrange's equation for  $r$ , with eq.(1) inserted, deduce that

$$\mu \ddot{r} - \frac{l^2}{\mu r^3} + \frac{dU(r)}{dr} = 0 \quad \dots\dots (2) \quad \text{(3 marks)}$$

- (iii) Multiply eq.(2) by  $dr$  and use  $\ddot{r} dr = \frac{d\dot{r}}{dt} dr = d\dot{r} \frac{dr}{dt} = \dot{r} d\dot{r} = d\left(\frac{\dot{r}^2}{2}\right)$  to

show that the total energy  $E (\equiv T + U)$  is conserved. (6 marks)

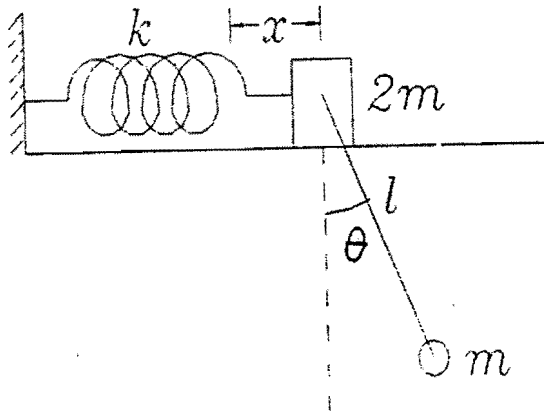
- (b) If an earth satellite of 100 kg mass is having a pure tangential speed

$$v_{\theta} (= r \dot{\theta}) = 10,000 \frac{m}{s} \quad \text{at its near-earth-point } 600 \text{ km above the earth surface,}$$

- (i) calculate the values of  $l$  and  $E$  of this satellite, (6 marks)  
 (ii) calculate the values of the eccentricity  $\varepsilon$  and show that the orbit is an elliptical orbit. Also calculate its period. (7 marks)

### Question four

A simple pendulum of mass  $m$  and length  $l$ , hangs from a supporting block of mass  $2m$  which can move along a horizontal line (in the plane of the pendulum), and is restricted by a spring with a spring constant  $k$  as shown below



- (i) For small  $\theta$ , i.e.,  $\left(\sin(\theta) \approx \theta \text{ and } \cos(\theta) \approx 1 - \frac{\theta^2}{2}\right)$ , show that the Lagrangian for the system can be expressed as:

$$L = \frac{3}{2} m \dot{x}^2 + \frac{1}{2} m l^2 \dot{\theta}^2 + m l \dot{x} \dot{\theta} - \frac{1}{2} m g l \theta^2 - \frac{1}{2} k x^2$$

where the zero gravitational potential is set at the equilibrium position. (6 marks)

- (ii) Write down the equations of motion and deduce that

$$\begin{cases} \ddot{x} = -\left(\frac{k}{2m}\right)x + \left(\frac{g}{2}\right)\theta \\ \ddot{\theta} = \left(\frac{k}{2ml}\right)x - \left(\frac{3g}{2l}\right)\theta \end{cases} \quad (10 \text{ marks})$$

- (iii) Set  $x = \hat{X}_1 e^{i\omega t}$  and  $\theta = \hat{X}_2 e^{i\omega t}$  (where  $\hat{X}_1$  and  $\hat{X}_2$  are constants) and deduce from the equations in (ii) the matrix equation  $-\omega^2 X = A X$  where

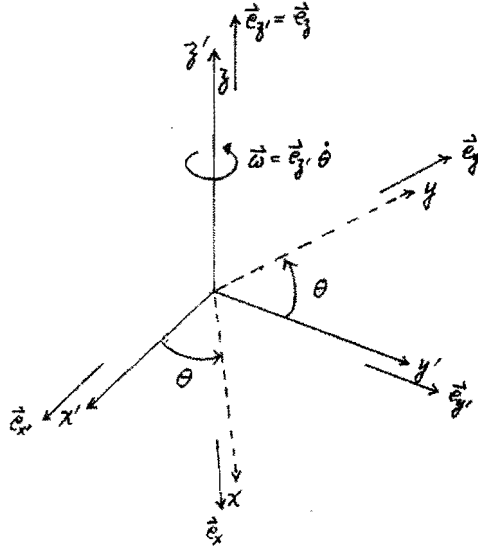
$$X = \begin{pmatrix} \hat{X}_1 \\ \hat{X}_2 \end{pmatrix} \text{ and } A = \begin{pmatrix} -\left(\frac{k}{2m}\right) & \left(\frac{g}{2}\right) \\ \frac{k}{2ml} & -\left(\frac{3g}{2l}\right) \end{pmatrix} \quad (4 \text{ marks})$$

- (iv) Show that the eigenfrequencies  $\omega$  of this coupled system satisfies the following equation

$$\omega^4 - \left(\frac{k}{2m} + \frac{3g}{2l}\right)\omega^2 + \left(\frac{k g}{2ml}\right) = 0 \quad (5 \text{ marks})$$

### Question five

- (a) Two set of Cartesian coordinate axes are having the same origins and z-axis. The non-prime system (referred to as “rotating” system) is rotating with an angular velocity  $\vec{\omega} = \vec{e}_z \dot{\theta}$  about the prime system (referred as “fixed” system) as shown below:



For any vector field  $\vec{F}$  decomposed into the above two-set of cartesian components, i.e.,  $\vec{F} = \vec{e}_x F_x + \vec{e}_y F_y + \vec{e}_z F_z = \vec{e}_{x'} F_{x'} + \vec{e}_{y'} F_{y'} + \vec{e}_{z'} F_{z'}$ , show that

$$\left( \frac{d\vec{F}}{dt} \right)_{fixed} = \left( \frac{d\vec{F}}{dt} \right)_{rotating} + \vec{\omega} \times \vec{F} \quad \text{where}$$

$$\left( \frac{d\vec{F}}{dt} \right)_{fixed} = \vec{e}_x \frac{dF_x}{dt} + \vec{e}_y \frac{dF_y}{dt} + \vec{e}_z \frac{dF_z}{dt} \quad \text{and}$$

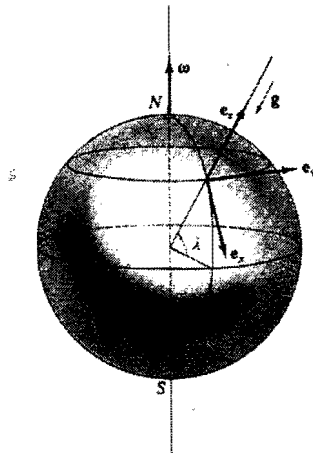
$$\left( \frac{d\vec{F}}{dt} \right)_{rotating} = \vec{e}_{x'} \frac{dF_{x'}}{dt} + \vec{e}_{y'} \frac{dF_{y'}}{dt} + \vec{e}_{z'} \frac{dF_{z'}}{dt}$$

( 12 marks )

(Hint :  $\vec{e}_x = \vec{e}_{x'} \cos(\theta) + \vec{e}_{y'} \sin(\theta)$  ,  $\vec{e}_y = -\vec{e}_{x'} \sin(\theta) + \vec{e}_{y'} \cos(\theta)$  and  $\vec{e}_z = \vec{e}_{z'}$ .)

Question five (continued)

(b)



If a particle is projected vertically upward with an initial speed  $v_0$  to a height  $h$  above a point on the earth's surface at northern latitude  $\lambda$ , show that it strikes the ground at a point  $\frac{4}{3} \omega \cos(\lambda) \sqrt{\frac{8 h^3}{g}}$  to the west. Neglect air resistance and only consider small vertical height. ( 13 marks )

(Hint :

$$\vec{a}_{eff} \approx \vec{e}_z (-g) - 2 \vec{\omega} \times \vec{v}_r, \quad \vec{v}_r \approx \vec{e}_z (v_0 - g t), \quad \vec{\omega} = \vec{e}_x (-\omega \cos(\lambda)) + \vec{e}_z (\omega \sin(\lambda))$$

$$\text{and } v_0 = \sqrt{2 g h}, \quad (\text{total time for the given motion}) = \frac{2 v_0}{g} )$$

**Useful informations**

$$V = - \int \vec{F} \cdot d\vec{l} \quad \text{and} \quad d\vec{l} = \vec{e}_x x + \vec{e}_y y + \vec{e}_z z \quad \text{in cartesian}$$

$$L = T - V = L(q_\alpha, \dot{q}_\alpha; t)$$

$$p_\alpha = \frac{\partial L}{\partial \dot{q}_\alpha}$$

$$\dot{p}_\alpha = \frac{\partial L}{\partial q_\alpha}$$

$$H = \sum_\alpha p_\alpha \dot{q}_\alpha - L = H(q_\alpha, p_\alpha; t) \xrightarrow{\text{if } \frac{\partial H}{\partial t} = 0} T + V$$

$$\dot{q}_\alpha = \frac{\partial H}{\partial p_\alpha}$$

$$\dot{p}_\alpha = - \frac{\partial H}{\partial q_\alpha}$$

$$[u, v] \equiv \sum_\alpha \left( \frac{\partial u}{\partial q_\alpha} \frac{\partial v}{\partial p_\alpha} - \frac{\partial u}{\partial p_\alpha} \frac{\partial v}{\partial q_\alpha} \right)$$

$$G = 6.673 \times 10^{-11} \frac{N m^2}{kg^2}$$

$$\text{radius of earth } r_E = 6.4 \times 10^6 \text{ m}$$

$$\text{mass of earth } m_E = 6 \times 10^{24} \text{ kg}$$

$$\text{earth attractive potential} \equiv - \frac{k}{r} \quad \text{where } k = G m m_E$$

$$\varepsilon = \sqrt{1 + \frac{2 E l^2}{\mu k^2}} \quad \{(\varepsilon = 0, \text{circle}), (0 < \varepsilon < 1, \text{ellipse}), (\varepsilon = 1, \text{parabola}), \dots\}$$

$$\mu = \frac{m_1 m_2}{m_1 + m_2} \approx m_1 \quad \text{if } m_2 \gg m_1$$

$$\text{For elliptical orbit, i.e., } 0 < \varepsilon < 1, \text{ then } \left\{ \begin{array}{l} \text{semi-major } a = \frac{k}{2|E|} \\ \text{semi-minor } b = \frac{l}{\sqrt{2\mu|E|}} \\ \text{period } \tau = \frac{2\mu}{l} (\pi a b) \end{array} \right.$$

$$I = \left( \begin{array}{ccc} \sum_\alpha m_\alpha (x_{\alpha,2}^2 + x_{\alpha,3}^2) & - \sum_\alpha m_\alpha x_{\alpha,1} x_{\alpha,2} & - \sum_\alpha m_\alpha x_{\alpha,1} x_{\alpha,3} \\ - \sum_\alpha m_\alpha x_{\alpha,2} x_{\alpha,1} & \sum_\alpha m_\alpha (x_{\alpha,1}^2 + x_{\alpha,3}^2) & - \sum_\alpha m_\alpha x_{\alpha,2} x_{\alpha,3} \\ - \sum_\alpha m_\alpha x_{\alpha,3} x_{\alpha,1} & - \sum_\alpha m_\alpha x_{\alpha,3} x_{\alpha,2} & \sum_\alpha m_\alpha (x_{\alpha,1}^2 + x_{\alpha,2}^2) \end{array} \right)$$



$$\vec{F}_{eff} = \vec{F} - m \ddot{\vec{R}}_f - m \dot{\vec{\omega}} \times \vec{r} - m \vec{\omega} \times (\vec{\omega} \times \vec{r}) - 2 m \vec{\omega} \times \vec{v}_r \quad \text{where}$$

$$\vec{r}' = \vec{R} + \vec{r} \quad \text{and}$$

$\vec{r}'$  refers to fixed (inertial system)

$\vec{r}$  refers to rotational (non-inertial system) rotates with  $\vec{\omega}$  to  $\vec{r}'$  system

$\vec{R}$  from the origin of  $\vec{r}'$  to the origin of  $\vec{r}$

$$\vec{v}_r = \left( \frac{d\vec{r}}{dt} \right)_r$$