UNIVERSITY OF SWAZILAND

## FACULTY OF SCIENCE AND ENGINEERING

## **DEPARTMENT OF PHYSICS**

MAIN EXAMINATION 2012/2013

TITLE OF PAPER	:	CLASSICAL MECHANICS
COURSE NUMBER	:	P320
TIME ALLOWED	:	THREE HOURS
INSTRUCTIONS	:	ANSWER <u>ANY FOUR</u> OUT OF FIVE QUESTIONS. EACH QUESTION CARRIES <u>25</u> MARKS. MARKS FOR DIFFERENT SECTIONS ARE SHOWN IN THE RIGHT-HAND MARGIN.

# THIS PAPER HAS <u>EIGHT</u> PAGES, INCLUDING THIS PAGE.

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#### P320 CLASSICAL MECHANICS

#### Question one

(a) Given the following definite integral  $J(\alpha) = \int_{x_1}^{x_2} f(y(\alpha, x), y'(\alpha, x); x) dx$ , where the varied integration path is  $y(\alpha, x) = y(x) + \alpha \eta(x)$  and  $\eta(x_1) = \eta(x_2) = 0$ as shown in the following diagram :



Using the extremum condition for  $J(\alpha)$ , i.e.,  $\frac{\partial J(\alpha)}{\partial \alpha}\Big|_{\alpha=0} = 0$ , to deduce that *f* along the extremum path i.e., f(v(x), v'(x); x), satisfies the following equation:

$$\frac{\partial f}{\partial y} - \frac{d}{dx} \left( \frac{\partial f}{\partial y'} \right) = 0 \quad . \tag{10 marks}$$

(b) If *H* denotes the Hamiltonian function and *L* is the Lagrangian function, use the definition  $H = \sum_{\alpha=1}^{n} p_{\alpha} \dot{q}_{\alpha} - L$  (where  $p_{\alpha}$  and  $q_{\alpha} (\alpha = 1, 2, \dots, n)$  are the generalized momenta and coordinates respectively, i.e.,  $H = H(q_1, \dots, q_n, p_1, \dots, p_n, t)$ ,  $L = L(q_1, \dots, q_n, \dot{q}_1, \dots, \dot{q}_n, t)$ ,  $p_{\alpha} = \frac{\partial L}{\partial \dot{q}_{\alpha}}$  and  $\dot{p}_{\alpha} = \frac{\partial L}{\partial q_{\alpha}}$ ) to show that (i)  $\dot{q}_{\alpha} = \frac{\partial H}{\partial p_{\alpha}}$   $\alpha = 1, 2, \dots, n$  (4 marks) (ii)  $\dot{p}_{\alpha} = -\frac{\partial H}{\partial q}$   $\alpha = 1, 2, \dots, n$  (4 marks)

(iii) 
$$\frac{\partial H}{\partial t} = -\frac{\partial L}{\partial t}$$
 (7 marks)

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#### **Ouestion two**

The Poisson Bracket [F,G] of two functions F and G of canonical variables (a)  $p_{\alpha}$  and  $q_{\alpha}$  is given by

$$\begin{bmatrix} F,G \end{bmatrix} = \sum_{\alpha=1}^{n} \left( \frac{\partial F}{\partial q_{\alpha}} \frac{\partial G}{\partial p_{\alpha}} - \frac{\partial F}{\partial p_{\alpha}} \frac{\partial G}{\partial q_{\alpha}} \right)$$
  
Show that  
(i)  $\begin{bmatrix} H,q_{\alpha} \end{bmatrix} = -\dot{q}_{\alpha}$  where  $H$  is the Hamiltonian, (4 marks)  
(ii)  $\begin{bmatrix} p_{\alpha}, p_{\beta} \end{bmatrix} = 0$  where  $\begin{cases} \alpha = 1, 2, \dots, n \\ \beta = 1, 2, \dots, n \end{cases}$ , (3 marks)

(b)

A pendulum is composed of a rigid rod of length b with a mass  $m_1$  at its end. Another mass  $m_2$  is placed halfway down the rod. The mass of the rod itself is negligible. Let the fixed and the body coordinate systems have their origin at the pendulum pivot point. Let  $(\vec{e}_{1}', \vec{e}_{2}', \vec{e}_{3}')$  and  $(\vec{e}_{1}, \vec{e}_{2}, \vec{e}_{3})$ be the unit vectors of the fixed and the body coordinate system respectively as shown below.



Write down the inertia tensor I for the pendulum with respect to the body (i) coordinate system given above and deduce that I is a diagonal matrix with its diagonal elements as  $I_{1,1} = 0$  and  $I_{2,2} = I_{3,3} = \left(m_1 + \frac{m_2}{4}\right)b^2$ . (6 marks)

From the equation of rotational motion, i.e.,  $\vec{L} = \vec{N}$  where angular momentum (ii)  $\vec{L} = I \vec{\omega}$  and torque  $\vec{N} = \sum_{\alpha} (\vec{r}_{\alpha} \times \vec{F}_{\alpha})$ , deduce the following equation :

$$b^{2}\left(m_{1}+\frac{m_{2}}{4}\right)\ddot{\theta}=-b\ g\ \sin(\theta)\left(m_{1}+\frac{m_{2}}{2}\right)$$
(12 marks)  
(Hint:

(Hint :

$$\vec{\omega} = \vec{e}_{3}' \dot{\theta} , \quad \vec{F}_{1} = \vec{e}_{1}' m_{1} g , \quad \vec{F}_{2} = \vec{e}_{1}' m_{2} g ,$$
  
$$\vec{r}_{1} = \vec{e}_{1}' b \cos(\theta) + \vec{e}_{2}' b \sin(\theta) \quad and \quad \vec{r}_{2} = \vec{e}_{1}' \frac{b}{2} \cos(\theta) + \vec{e}_{2}' \frac{b}{2} \sin(\theta) )$$

#### **Question three**

(a) For circular orbits in an attractive central force potential of the form  $V = -\frac{k}{r^n}$ 

where k is a positive constant and n > 0, find a relation between the kinetic and potential energies and show that

$$T = \frac{n\,k}{2\,r^n} \qquad (9 \text{ marks})$$

(Hint:  $\vec{a} = \vec{e}_r \left( \ddot{r} - r \dot{\theta}^2 \right) + \vec{e}_{\theta} \left( 2 \dot{r} \dot{\theta} + r \ddot{\theta} \right)$ )

(b) Starting from the law of conservation of angular momentum l, derive Kepler's third law, i.e., the relation between the period  $\tau$  of a closed orbit in an attractive inverse square central force and the area A of the orbit. Show that

$$\tau = \frac{2 \mu}{l} A$$
 where  $\mu$  is the reduced mass of the system. (7 marks)

(c) An earth satellite moves in an elliptical orbit with period  $\tau$ , eccentricity  $\varepsilon$  and semi-major axis a. The maximum radial velocity, named as  $v_{\theta, \max}$ , occurs at  $r = r_{\min}$ . Show that

$$v_{\theta,\max} = \frac{2 \pi a}{\tau \sqrt{1 - \varepsilon^2}}$$
 (9 marks)

(Hint:  $\mu r_{\min} v_{\theta, \max} = l$ ,  $A = \pi a b$  and  $b = a \sqrt{1 - \varepsilon^2}$ )

### **Question four**

Two pendulums of equal lengths b and equal masses m are connected by a spring of force constant k as shown below. The spring is unstretched in the equilibrium position, i.e.,  $\theta_1 = 0$  and  $\theta_2 = 0$ .



(i) For small 
$$\theta_1$$
 and  $\theta_2$ , i.e.,  

$$\begin{pmatrix} \sin(\theta_1) \approx \theta_1 & \sin(\theta_2) \approx \theta_2 & \cos(\theta_1) \approx 1 - \frac{\theta_1^2}{2} & and & \cos(\theta_2) \approx 1 - \frac{\theta_2^2}{2} \end{pmatrix}$$
, show that the

Lagrangian for the system can be expressed as:

(ii)

$$L = \frac{1}{2} m b^{2} \left( \dot{\theta}_{1}^{2} + \dot{\theta}_{2}^{2} \right) - \frac{m g b}{2} \left( \theta_{1}^{2} + \theta_{2}^{2} \right) - \frac{k b^{2}}{2} \left( \theta_{1} - \theta_{2} \right)^{2}$$

where the zero gravitational potential is set at the equilibrium position. (8 marks) Write down the equations of motion and deduce that

$$\begin{cases} \ddot{\theta}_{1} = -\left(\frac{m \ g + k \ b}{m \ b}\right)\theta_{1} + \frac{k}{m} \ \theta_{2} \\ \ddot{\theta}_{2} = \frac{k}{m} \theta_{1} - \left(\frac{m \ g + k \ b}{m \ b}\right)\theta_{2} \end{cases}$$
(6 marks)

(iii) Set  $\theta_1 = \hat{X}_1 e^{i\omega t}$  and  $\theta_2 = \hat{X}_2 e^{i\omega t}$  (where  $\hat{X}_1$  and  $\hat{X}_2$  are constants) and deduce from the equations in (ii) the matrix equation  $-\omega^2 X = A X$  where

$$X = \begin{pmatrix} \hat{X}_1 \\ \hat{X}_2 \end{pmatrix} \text{ and } A = \begin{pmatrix} -\begin{pmatrix} m g + k b \\ m \end{pmatrix} & \frac{k b}{m} \\ \frac{k b}{m} & -\begin{pmatrix} m g + k b \\ m \end{pmatrix} \end{pmatrix}$$
(4 marks)

(iv) Find the eigenfrequencies  $\omega$  of this coupled system and show that they are

$$\omega = \sqrt{\frac{g}{b}} \qquad or \qquad \sqrt{\frac{m \, g + 2 \, k \, b}{m \, b}} \tag{7 marks}$$

#### **Question five**

(a) Two sets of coordinate systems are having the same origins. The non-prime system (with position vector denoted as  $\vec{r}$  and referred to as "rotating" system) is rotating with an angular velocity  $\vec{\omega}$  about the prime system (with position vector denoted as  $\vec{r}'$  and referred as "fixed" system and taken as an inertial system). Use the following proven

relation that  $\left(\frac{d\vec{F}}{dt}\right)_{fixed} = \left(\frac{d\vec{F}}{dt}\right)_{rotating} + \vec{\omega} \times \vec{F}$  for any vector field  $\vec{F}$  to deduce the

following:

$$\vec{a}_{eff} = \vec{a} - \vec{\omega} \times \vec{r} - \vec{\omega} \times (\vec{\omega} \times \vec{r}) - 2 \vec{\omega} \times \vec{v}_r \quad \text{where}$$
  
$$\vec{r}' = \vec{r} \quad (\text{same origin}) \quad , \quad \vec{a}_{eff} \equiv \left(\frac{d^2 \vec{r}}{dt^2}\right)_{\text{rotating}} \quad , \quad \vec{a} \equiv \left(\frac{d^2 \vec{r}'}{dt^2}\right)_{\text{fixed}} \quad , \quad \vec{v}_r \equiv \left(\frac{d \vec{r}}{dt}\right)_{\text{rotating}} ,$$
  
$$\left(\frac{d \vec{v}_r}{dt}\right)_{\text{rotating}} \equiv \left(\frac{d^2 \vec{r}}{dt^2}\right)_{\text{rotating}} \quad \text{and} \quad \vec{\omega} \equiv \left(\frac{d \vec{\omega}}{dt}\right)_{\text{rotating}}$$
(12 marks )

(b)



Show that the horizontal deflection d along  $-\vec{e}_{y}$  direction resulting from the Coriolis force  $(-2 \ m \ \vec{\omega} \times \vec{v}_{r})$  of a particle falling freely in the earth's gravitational field at a northern latitude  $\lambda$  is

$$d \approx \frac{1}{3} \omega \cos(\lambda) \sqrt{\frac{8 h^3}{g}}$$
 where

 $\omega : angular velocity of earth's rotation$ (13 marks) h : the height of the particle above the earth before its free fall(Hint: $<math>\vec{a}_{eff} \approx \vec{e}_z (-g) - 2 \vec{\omega} \times \vec{v}_r$ ,  $\vec{v}_r \approx \vec{e}_z (-gt)$ ,  $\vec{\omega} = \vec{e}_x (-\omega \cos(\lambda)) + \vec{e}_z (\omega \sin(\lambda))$ and (total time for the given motion) =  $\sqrt{\frac{2h}{g}}$ ) đ

$$\begin{split} V &= -\int \vec{F} \cdot d\vec{l} \quad \text{and reversely} \quad \vec{F} = -\vec{\nabla} V \\ L &= T - V = L(q_1, q_2, \cdots, q_n, \dot{q}_1, \dot{q}_2, \cdots, \dot{q}_n, t) \\ p_\alpha &= \frac{\partial L}{\partial \dot{q}_\alpha} \quad \text{and} \quad \dot{p}_\alpha = \frac{\partial L}{\partial q_\alpha} \\ H &= \sum_{\alpha=1}^n (p_\alpha \dot{q}_\alpha) - L = H(q_1, q_2, \cdots, q_n, \dot{q}_1, \dot{q}_2, \cdots, \dot{q}_n, t) \\ \dot{q}_\alpha &= \frac{\partial H}{\partial p_\alpha} \quad \text{and} \quad \dot{p}_\alpha = -\frac{\partial H}{\partial q_\alpha} \\ [u, v] &= \sum_{\alpha=1}^n \left( \frac{\partial u}{\partial q_\alpha} \frac{\partial v}{\partial p_\alpha} - \frac{\partial u}{\partial p_\alpha} \frac{\partial v}{\partial q_\alpha} \right) \\ G &= 6.673 \times 10^{-11} \quad \frac{N m^2}{kg^2} \\ \text{radius of earth} \quad r_E = 6.4 \times 10^6 \quad m \\ \text{mass of earth} \quad m_E = 6 \times 10^{24} \quad kg \\ \text{earth attractive potential} &= -\frac{k}{r} \quad \text{where} \quad k = G m \, m_E \\ \varepsilon &= \sqrt{1 + \frac{2 E l^2}{\mu \, k}} \quad \{(\varepsilon = 0, \text{circle}), (0 < \varepsilon < 1, \text{ellipse}), (\varepsilon = 1, \text{parabola}), \cdots\} \\ \mu &= \frac{m_1 m_2}{m_1 + m_2} \approx m_1 \quad \text{if} \quad m_2 \gg m_1 \\ For \text{ elliptical orbit, i.e., } 0 < \varepsilon < 1, \text{ then} \begin{cases} \text{semi-major } a = \frac{k}{2|E|} \\ \text{semi-min or } b = \frac{l}{\sqrt{2 \mu |E|}} \\ \text{period } \tau = \frac{2 \mu}{l} (\pi \, a \, b) \\ r_{\min} = a (1 - \varepsilon) \quad \& r_{\max} = a (1 + \varepsilon) \end{cases}$$

for plane polar  $(r, \theta)$  system with unit vectors  $(\vec{e}_r, \vec{e}_{\theta})$ , we have  $\begin{cases} \vec{v} = \vec{e}_r \ \dot{r} + \vec{e}_{\theta} \ r \ \dot{\theta} \\ \vec{a} = \vec{e}_r \ (\vec{r} - r \ \dot{\theta}^2) + \vec{e}_{\theta} \ (2 \ \dot{r} \ \dot{\theta} + r \ \ddot{\theta}) \end{cases}$   $\vec{\nabla} f = \vec{e}_r \ \frac{\partial f}{\partial r} + \vec{e}_{\theta} \ \frac{1}{r} \ \frac{\partial f}{\partial \theta}$ 

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## Useful informations (continued)

$$I = \begin{pmatrix} \sum_{\alpha} m_{\alpha} \left( x_{\alpha,2}^{2} + x_{\alpha,3}^{2} \right) & -\sum_{\alpha} m_{\alpha} x_{\alpha,1} x_{\alpha,2} & -\sum_{\alpha} m_{\alpha} x_{\alpha,1} x_{\alpha,3} \\ -\sum_{\alpha} m_{\alpha} x_{\alpha,2} x_{\alpha,1} & \sum_{\alpha} m_{\alpha} \left( x_{\alpha,1}^{2} + x_{\alpha,3}^{2} \right) & -\sum_{\alpha} m_{\alpha} x_{\alpha,2} x_{\alpha,3} \\ -\sum_{\alpha} m_{\alpha} x_{\alpha,3} x_{\alpha,1} & -\sum_{\alpha} m_{\alpha} x_{\alpha,3} x_{\alpha,2} & \sum_{\alpha} m_{\alpha} \left( x_{\alpha,1}^{2} + x_{\alpha,2}^{2} \right) \end{pmatrix}$$

 $\vec{F}_{eff} = \vec{F} - m \, \vec{R}_f - m \, \vec{\omega} \times \vec{r} - m \, \vec{\omega} \times (\vec{\omega} \times \vec{r}) - 2 \, m \, \vec{\omega} \times \vec{v}_r \qquad \text{where}$  $\vec{r}' = \vec{R} + \vec{r} \quad \text{and}$  $\vec{r}' \quad refers \ to \quad fixed (inertial \ system)$ 

- $\vec{r}$  refers to rotatinal (non-inertial system) rotates with  $\vec{\omega}$  to  $\vec{r}$ ' system
- $\vec{R}$  from the origin of  $\vec{r}$ ' to the origin of  $\vec{r}$

$$\vec{v}_r = \left(\frac{d\vec{r}}{dt}\right)_r$$