

UNIVERSITY OF SWAZILAND

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FACULTY OF SCIENCE AND ENGINEERING

DEPARTMENT OF PHYSICS

SUPPLEMENTARY EXAMINATION 2013/2014

TITLE OF PAPER : CLASSICAL MECHANICS

COURSE NUMBER : P320

TIME ALLOWED : THREE HOURS

**INSTRUCTIONS : ANSWER ANY FOUR OUT OF FIVE
QUESTIONS.
EACH QUESTION CARRIES 25
MARKS.
MARKS FOR DIFFERENT SECTIONS
ARE SHOWN IN THE RIGHT-HAND
MARGIN.**

THIS PAPER HAS TWELVE PAGES, INCLUDING THIS PAGE.

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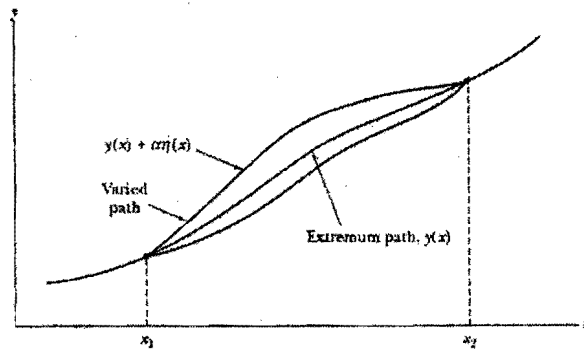
Question one

(a) Given the following definite integral of

$$J(\alpha) = \int_{x_1}^{x_2} f(y(\alpha, x), y'(\alpha, x), y''(\alpha, x), y'''(\alpha, x); x) dx, \text{ where the varied integration path is } y(\alpha, x) = y(x) + \alpha \eta(x), \quad \eta(x_1) = \eta(x_2) = 0,$$

$$\left. \frac{d\eta(x)}{dx} \right|_{x=x_1} = \left. \frac{d\eta(x)}{dx} \right|_{x=x_2} = 0 \quad \& \quad \left. \frac{d^2\eta(x)}{dx^2} \right|_{x=x_1} = \left. \frac{d^2\eta(x)}{dx^2} \right|_{x=x_2} = 0 \quad \text{as shown in the}$$

following diagram :

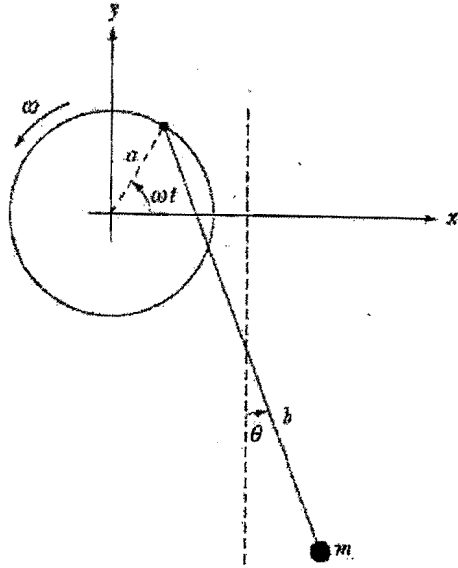


Using the extremum condition for $J(\alpha)$, i.e., $\left. \frac{\partial J(\alpha)}{\partial \alpha} \right|_{\alpha=0} = 0$, to deduce that

f along the extremum path, i.e., $f(y(x), y'(x), y''(x), y'''(x); x)$, satisfies the following equation:

$$\frac{\partial f}{\partial y} - \frac{d}{dx} \left(\frac{\partial f}{\partial y'} \right) + \frac{d^2}{dx^2} \left(\frac{\partial f}{\partial y''} \right) - \frac{d^3}{dx^3} \left(\frac{\partial f}{\partial y'''} \right) = 0 \quad (12 \text{ marks})$$

- (b) A simple pendulum of length b and mass m moves on a mass-less rim of radius a rotating with constant angular velocity ω as shown in the figure below:



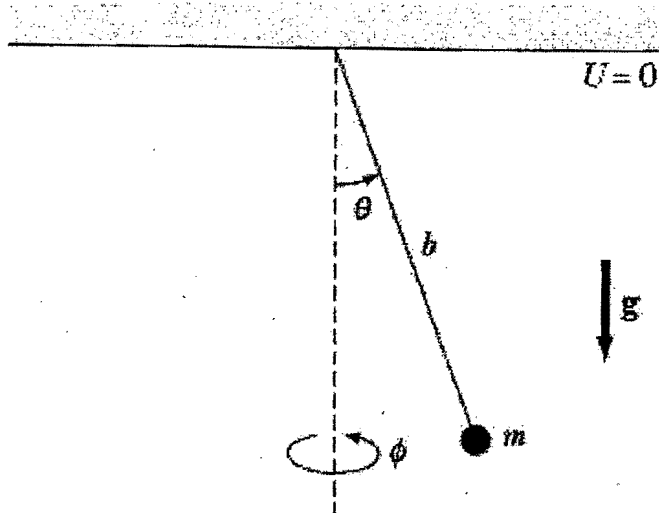
Write down the Lagrangian of the system in terms of θ and then deduce the following equation of motion

$$\ddot{\theta} - \frac{a}{b} \omega^2 \cos(\theta - \omega t) + \frac{g}{b} \sin(\theta) = 0 \quad (13 \text{ marks})$$

Question two

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A spherical pendulum of mass m and length b is shown in the figure below:



- (a) (i) From $x = b \sin(\theta) \cos(\phi)$, $y = b \sin(\theta) \sin(\phi)$ & $z = -b \cos(\theta)$ and $T = \frac{1}{2} m (\dot{x}^2 + \dot{y}^2 + \dot{z}^2)$ & $V = m g z$, deduce the following Lagrangian for the system in terms of θ & ϕ as
- $$L = \frac{1}{2} m b^2 (\dot{\theta}^2 + \dot{\phi}^2 \sin^2(\theta)) + m g b \cos(\theta) \dots\dots (1) \quad (5 \text{ marks})$$
- (ii) Write down the equations of motion for θ & ϕ and deduce that
- $$\begin{cases} \frac{d p_{\theta}}{d t} = m b^2 \sin(\theta) \cos(\theta) \dot{\phi}^2 - m g b \sin(\theta) \dots\dots (2) \\ \frac{d p_{\phi}}{d t} = 0 \dots\dots (3) \end{cases}$$
- where $p_{\theta} = m b^2 \dot{\theta}$ & $p_{\phi} = m b^2 \sin^2(\theta) \dot{\phi}$
- (5 marks)
- (iii) From eq.(3), one has $p_{\phi} = \text{const.} \xrightarrow{\text{set as}} K$, deduce from eq.(2) the following equation for small θ , i.e., $\left(\sin(\theta) \approx \theta \text{ and } \cos(\theta) \approx 1 - \frac{\theta^2}{2} \text{ or } 1 \right)$, that
- $$m^2 b^4 \theta^3 \ddot{\theta} = K^2 - m^2 g b^3 \theta^4 \dots\dots (4) \quad (4 \text{ marks})$$
- (iv) If $K = 0$ in eq.(4), write down the general solution of $\theta(t)$. (3 marks)
- (b) (i) Find the Hamiltonian of the system in terms of θ , ϕ , p_{θ} & p_{ϕ} . (4 marks)
- (ii) Write down the equations of motion for H in (b)(i). (4 marks)

Question three

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(a) Given the Lagrangian for the two-body central force system as :

$$L = T - V = \frac{1}{2} \mu (\dot{r}^2 + r^2 \dot{\theta}^2) + \frac{k}{r}$$

where μ is the reduced mass of the system, k is a positive constant and (r, θ) are polar coordinates of the motion plane with its origin at the center of mass of the two-body system. The integral form of orbital equation can be written as

$$\theta = \int \frac{\left(\frac{l}{r^2}\right)}{\sqrt{2\mu \left(E - \frac{l^2}{2\mu r^2} + \frac{k}{r}\right)}} dr + const. \quad \dots\dots (1)$$

where $l = \mu r^2 \dot{\theta}$ (i.e., angular momentum) and $E = \frac{1}{2} \mu (\dot{r}^2 + r^2 \dot{\theta}^2) - \frac{k}{r}$ (i.e., total energy) are two constants of the system.

Choose the integration constant in eq.(1) as zero (i.e., this is the same as choosing the initial value of r as r_{\min} at $\theta = 0$), and set $u \equiv \frac{1}{r}$,

(i) show that eq.(1) can be simplified as

$$\theta = - \int \frac{1}{\sqrt{\left(\frac{2\mu E}{l^2} - u^2 + \frac{2\mu k}{l^2} u\right)}} du \quad \dots\dots (2) \quad (2 \text{ marks})$$

(ii) combine $\left(u^2 - \frac{2\mu k}{l^2} u\right)$ in eq.(2) into the first two terms of a perfect square of u' and show that eq.(2) can be further simplified to

$$\theta = - \int \frac{1}{\sqrt{(a^2 - (u')^2)}} du' \quad \dots\dots (3) \quad (2 \text{ marks})$$

where $a = \sqrt{\frac{\mu^2 k^2}{l^4} + \frac{2\mu E}{l^2}}$ & $u' \equiv u - \frac{\mu k}{l^2}$

(iii) set $u' = a \cos(\beta)$ and carry out the integral of $\int \frac{1}{\sqrt{(a^2 - (u')^2)}} du'$ and show that eq.(3) becomes $\theta = \beta \dots\dots (4) \quad (2 \text{ marks})$

(iv) Taking cosine of eq.(4) and using $u' = a \cos(\beta)$, $u' \equiv u - \frac{\mu k}{l^2}$ & $u = \frac{1}{r}$, deduce the following orbital equation

$$\frac{\alpha}{r} \equiv 1 + \varepsilon \cos(\theta) \quad \text{where} \quad \alpha \equiv \frac{l^2}{\mu k} \quad \& \quad \varepsilon \equiv \sqrt{1 + \frac{2 E l^2}{\mu k^2}} \quad (6 \text{ marks})$$

Question three (continued)

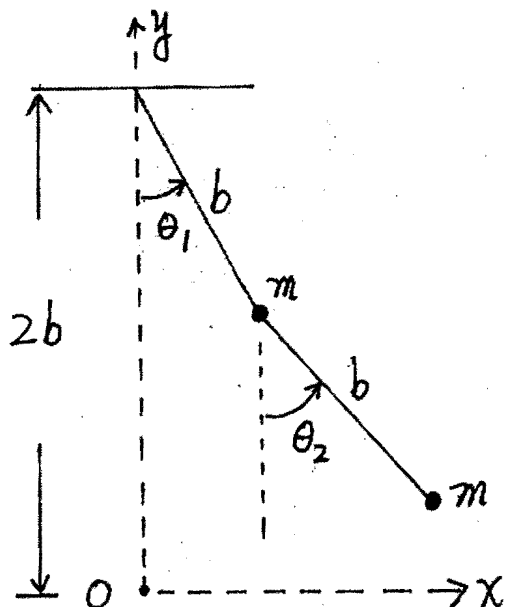
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- (b) If an earth satellite of 500 kg mass is having a pure tangential speed $v_\theta (= r \dot{\theta}) = 9,000 \frac{\text{m}}{\text{s}}$ at its near-earth-point 800 km above the earth surface,
- (i) calculate the values of l and E of this satellite, (4 marks)
 - (ii) calculate the values of the eccentricity ε and show that the orbit is an elliptical orbit. Also calculate its period. (2+4 marks)
 - (iii) what should be the minimum value of the v_θ at the same given near-earth-point such that the satellite would have a open orbit? (3 marks)

Question four

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Consider the motion of the bobs in the double pendulum system in the figure below.



Both pendulums are identical and having the length b and bob mass m . The motion of both bobs is restricted to lie in the plane of this paper, i.e., x-y plane.

- (a) (i) For small θ_1 and θ_2 , i.e., $\left(\sin(\theta) \approx \theta \text{ and } \cos(\theta) \approx 1 - \frac{\theta^2}{2} \text{ or } 1 \right)$, show that the Lagrangian for the system can be expressed as:

$$L = m b^2 \dot{\theta}_1^2 + \frac{1}{2} m b^2 \dot{\theta}_2^2 + m b^2 \dot{\theta}_1 \dot{\theta}_2 - m g b \left(1 + \theta_1^2 + \frac{\theta_2^2}{2} \right) \dots\dots (1)$$

where the zero gravitational potential is set at the equilibrium position of the lower bob, i.e., $\theta_1 = 0$, $\theta_2 = 0$ and $y = 0$. (5 marks)

- (ii) Write down the equations of motion and deduce that

$$\begin{cases} 2 \ddot{\theta}_1 + \ddot{\theta}_2 = -2 \frac{g}{b} \theta_1 & \dots\dots (2) \\ \ddot{\theta}_1 + \ddot{\theta}_2 = -\frac{g}{b} \theta_2 & \dots\dots (3) \end{cases} \quad (5 \text{ marks})$$

- (iii) Deduce from eq.(2) & eq.(3) the following :

$$\begin{cases} \ddot{\theta}_1 = -2 \frac{g}{b} \theta_1 + \frac{g}{b} \theta_2 & \dots\dots (4) \\ \ddot{\theta}_2 = 2 \frac{g}{b} \theta_1 - 2 \frac{g}{b} \theta_2 & \dots\dots (5) \end{cases} \quad (3 \text{ marks})$$

Question four (continued)

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- (b) (i) Set $\theta_1 = \hat{X}_1 e^{i\omega t}$ and $\theta_2 = \hat{X}_2 e^{i\omega t}$ (where \hat{X}_1 and \hat{X}_2 are constants) and deduce from eq.(4) & eq.(5) the matrix equation $-\omega^2 X = A X$ where

$$X = \begin{pmatrix} \hat{X}_1 \\ \hat{X}_2 \end{pmatrix} \quad \text{and} \quad A = \begin{pmatrix} -\left(2\frac{g}{b}\right) & \frac{g}{b} \\ 2\frac{g}{b} & -\left(2\frac{g}{b}\right) \end{pmatrix} \quad (3 \text{ marks})$$

- (ii) Find the eigenfrequencies ω of this coupled system and show that they are

$$\omega_1 = \sqrt{\left(2 - \sqrt{2}\right)\frac{g}{b}} \quad \& \quad \omega_2 = \sqrt{\left(2 + \sqrt{2}\right)\frac{g}{b}} \quad (5 \text{ marks})$$

- (iii) Find the eigenvector corresponding to ω_1 in (b)(ii). (4 marks)

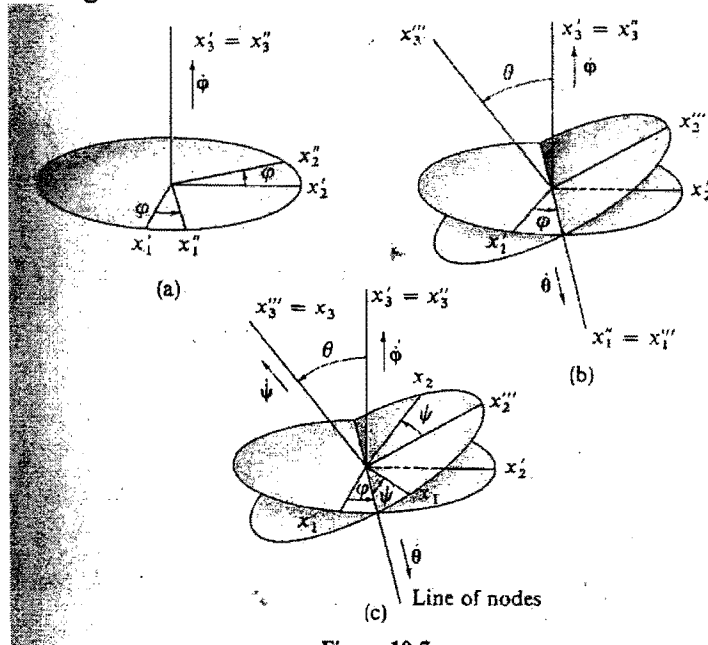
Question five

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- (a) The fixed (or inertia) coordinate system X' shares the same origin with the body coordinate system X such that only rotational motion is considered. The rotational velocity $\vec{\omega}$ of the body system with respect to the fixed system are breaking down into three independent angular velocities, i.e., $\vec{\omega} = \vec{\phi} + \vec{\theta} + \vec{\psi}$ where (ϕ, θ, ψ) are Eulerian angles. We use two intermediate coordinate systems X'' & X''' to bridge between X' & X systems such that $X'' = \lambda_\phi X'$, $X''' = \lambda_\theta X''$ & $X = \lambda_\psi X'''$ where

$$\lambda_\phi = \begin{pmatrix} \cos(\phi) & \sin(\phi) & 0 \\ -\sin(\phi) & \cos(\phi) & 0 \\ 0 & 0 & 1 \end{pmatrix}, \lambda_\theta = \begin{pmatrix} 1 & 0 & 0 \\ 0 & \cos(\theta) & \sin(\theta) \\ 0 & -\sin(\theta) & \cos(\theta) \end{pmatrix}, \lambda_\psi = \begin{pmatrix} \cos(\psi) & \sin(\psi) & 0 \\ -\sin(\psi) & \cos(\psi) & 0 \\ 0 & 0 & 1 \end{pmatrix}$$

as shown in the figure below.



- (i) Since the direction of $\vec{\phi}$ is along x_3' -axis (which is the same as x_3''' -axis) with

the magnitude of $\dot{\phi}$ thus $(\vec{\phi})' = \begin{pmatrix} 0 \\ 0 \\ \dot{\phi} \end{pmatrix}$ in X'' system, show that $\vec{\phi}$ in X

system (i.e., the body system) is $(\vec{\phi}) = \begin{pmatrix} \dot{\phi} \sin(\theta) \sin(\psi) \\ \dot{\phi} \sin(\theta) \cos(\psi) \\ \dot{\phi} \cos(\theta) \end{pmatrix}$ in X system. In other

words, show that

$$\begin{pmatrix} \dot{\phi} \sin(\theta) \sin(\psi) \\ \dot{\phi} \sin(\theta) \cos(\psi) \\ \dot{\phi} \cos(\theta) \end{pmatrix} = \lambda_\psi \lambda_\theta \begin{pmatrix} 0 \\ 0 \\ \dot{\phi} \end{pmatrix}$$

(5 marks)

- (ii) Since the direction of $\vec{\theta}$ is along x_1'' -axis (which is the same as x_1''' -axis) with the magnitude of $\dot{\theta}$ thus $\begin{pmatrix} \dot{\theta} \\ 0 \\ 0 \end{pmatrix}$ in X''' system, show that $\vec{\theta}$ in X system (i.e., the body system) is $\begin{pmatrix} \dot{\theta} \cos(\psi) \\ -\dot{\theta} \sin(\psi) \\ 0 \end{pmatrix}$. In other words, show that

$$\begin{pmatrix} \dot{\theta} \cos(\psi) \\ -\dot{\theta} \sin(\psi) \\ 0 \end{pmatrix} = \lambda_\psi \begin{pmatrix} \dot{\theta} \\ 0 \\ 0 \end{pmatrix} \quad (3 \text{ marks})$$

- (iii) Since any rotational velocity of a rigid body can be expressed as $\vec{\omega} = \vec{\phi} + \vec{\theta} + \vec{\psi}$ deduce that $\vec{\omega}$ in X system (i.e., body system) in terms of Eulerian angles is

$$\begin{pmatrix} \omega_1 \\ \omega_2 \\ \omega_3 \end{pmatrix} = \begin{pmatrix} \dot{\phi} \sin(\theta) \sin(\psi) + \dot{\theta} \cos(\psi) \\ \dot{\phi} \sin(\theta) \cos(\psi) - \dot{\theta} \sin(\psi) \\ \dot{\phi} \cos(\theta) + \dot{\psi} \end{pmatrix} \text{ in } X \text{ system} \quad (4 \text{ marks})$$

- (b) (i) By proper choice of the orientation of the body coordinate system, the inertia tensor I (i.e., rotational mass) of a rigid body can be in the form of a diagonalized

matrix, i.e., $I = \begin{pmatrix} I_1 & 0 & 0 \\ 0 & I_2 & 0 \\ 0 & 0 & I_3 \end{pmatrix}$, thus its rotational kinetic is

$$T_{rot} = \frac{1}{2} I_1 \omega_1^2 + \frac{1}{2} I_2 \omega_2^2 + \frac{1}{2} I_3 \omega_3^2.$$

Consider a torque free pure rotational motion of the rigid body, then its Lagrangian is

$$L = T_{rot} = \frac{1}{2} I_1 \omega_1^2 + \frac{1}{2} I_2 \omega_2^2 + \frac{1}{2} I_3 \omega_3^2 \rightarrow L(\phi, \theta, \psi, \dot{\phi}, \dot{\theta}, \dot{\psi}).$$

Write down the Lagrange equation of motion for ψ and deduce that

$$(I_1 - I_2) \omega_1 \omega_2 - I_3 \dot{\omega}_3 = 0 \dots \dots (1) \quad (11 \text{ marks})$$

- (ii) Based on what argument one can write down the other two equations of motion directly from eq.(1) in (b)(i) as

$$(I_2 - I_3) \omega_2 \omega_3 - I_1 \dot{\omega}_1 = 0 \quad \& \quad (I_3 - I_1) \omega_3 \omega_1 - I_2 \dot{\omega}_2 = 0$$

without going through the similar process of finding the equations of motion for the other two Eulerian angles? (2 marks)

$$V = - \int \vec{F} \cdot d\vec{l} \quad \text{and reversely} \quad \vec{F} = -\vec{\nabla} V$$

$$L = T - V = L(q_1, q_2, \dots, q_n, \dot{q}_1, \dot{q}_2, \dots, \dot{q}_n, t)$$

$$p_\alpha = \frac{\partial L}{\partial \dot{q}_\alpha} \quad \text{and} \quad \dot{p}_\alpha = \frac{\partial L}{\partial q_\alpha}$$

$$H = \sum_{\alpha=1}^n (p_\alpha \dot{q}_\alpha) - L = H(q_1, q_2, \dots, q_n, \dot{q}_1, \dot{q}_2, \dots, \dot{q}_n, t)$$

$$\dot{q}_\alpha = \frac{\partial H}{\partial p_\alpha} \quad \text{and} \quad \dot{p}_\alpha = -\frac{\partial H}{\partial q_\alpha}$$

$$[u, v] \equiv \sum_{\alpha=1}^n \left(\frac{\partial u}{\partial q_\alpha} \frac{\partial v}{\partial p_\alpha} - \frac{\partial u}{\partial p_\alpha} \frac{\partial v}{\partial q_\alpha} \right)$$

$$G = 6.673 \times 10^{-11} \frac{N m^2}{kg^2}$$

$$\text{radius of earth } r_E = 6.4 \times 10^6 \text{ m}$$

$$\text{mass of earth } m_E = 6 \times 10^{24} \text{ kg}$$

$$\text{earth attractive potential} \equiv -\frac{k}{r} \quad \text{where } k = G m m_E$$

$$\varepsilon = \sqrt{1 + \frac{2 E l^2}{\mu k}} \quad \{(\varepsilon = 0, \text{circle}), (0 < \varepsilon < 1, \text{ellipse}), (\varepsilon = 1, \text{parabola}), \dots\}$$

$$\mu = \frac{m_1 m_2}{m_1 + m_2} \approx m_1 \quad \text{if } m_2 \gg m_1$$

$$\text{For elliptical orbit, i.e., } 0 < \varepsilon < 1, \text{ then } \left\{ \begin{array}{l} \text{semi-major } a = \frac{k}{2|E|} \\ \text{semi-minor } b = \frac{l}{\sqrt{2\mu|E|}} \\ \text{period } \tau = \frac{2\mu}{l} (\pi a b) \\ r_{\min} = a(1 - \varepsilon) \text{ \& } r_{\max} = a(1 + \varepsilon) \end{array} \right.$$

for plane polar (r, θ) system with unit vectors $(\vec{e}_r, \vec{e}_\theta)$, we have

$$\left\{ \begin{array}{l} \vec{v} = \vec{e}_r \dot{r} + \vec{e}_\theta r \dot{\theta} \\ \vec{a} = \vec{e}_r (\ddot{r} - r \dot{\theta}^2) + \vec{e}_\theta (2\dot{r} \dot{\theta} + r \ddot{\theta}) \end{array} \right.$$

$$\vec{\nabla} f = \vec{e}_r \frac{\partial f}{\partial r} + \vec{e}_\theta \frac{1}{r} \frac{\partial f}{\partial \theta}$$

$$I = \begin{pmatrix} \sum_{\alpha} m_{\alpha} (x_{\alpha,2}^2 + x_{\alpha,3}^2) & -\sum_{\alpha} m_{\alpha} x_{\alpha,1} x_{\alpha,2} & -\sum_{\alpha} m_{\alpha} x_{\alpha,1} x_{\alpha,3} \\ -\sum_{\alpha} m_{\alpha} x_{\alpha,2} x_{\alpha,1} & \sum_{\alpha} m_{\alpha} (x_{\alpha,1}^2 + x_{\alpha,3}^2) & -\sum_{\alpha} m_{\alpha} x_{\alpha,2} x_{\alpha,3} \\ -\sum_{\alpha} m_{\alpha} x_{\alpha,3} x_{\alpha,1} & -\sum_{\alpha} m_{\alpha} x_{\alpha,3} x_{\alpha,2} & \sum_{\alpha} m_{\alpha} (x_{\alpha,1}^2 + x_{\alpha,2}^2) \end{pmatrix}$$

$$\vec{F}_{eff} = \vec{F} - m \ddot{\vec{R}}_f - m \dot{\vec{\omega}} \times \vec{r} - m \vec{\omega} \times (\vec{\omega} \times \vec{r}) - 2 m \vec{\omega} \times \vec{v}_r \quad \text{where}$$

$$\vec{r}' = \vec{R} + \vec{r} \quad \text{and}$$

\vec{r}' refers to fixed (inertial system)

\vec{r} refers to rotatinal (non-inertial system) rotates with $\vec{\omega}$ to \vec{r}' system

\vec{R} from the origin of \vec{r}' to the origin of \vec{r}

$$\vec{v}_r = \left(\frac{d\vec{r}}{dt} \right)_r$$