UNIVERSITY OF SWAZILAND

FACULTY OF SCIENCE

,

з.

DEPARTMENT OF PHYSICS

MAIN EXAMINATION 2016/2017

TITLE OF PAPER:CLASSICAL MECHANICSCOURSE NUMBER:P320TIME ALLOWED:THREE HOURS

INSTRUCTIONS : ANSWER <u>ANY FOUR</u> OUT OF FIVE QUESTIONS. EACH QUESTION CARRIES <u>25</u> MARKS. MARKS FOR DIFFERENT SECTIONS ARE SHOWN IN THE RIGHT-HAND MARGIN.

83

THIS PAPER HAS <u>TEN</u> PAGES, INCLUDING THIS PAGE.

DO NOT OPEN THE PAPER UNTIL PERMISSION HAS BEEN GIVEN BY THE INVIGILATOR.

1

• 4 •

P320 CLASSICAL MECHANICS

3

Question one

If H denotes the Hamiltonian function and L is the Lagrangian function, use the (a) definition $H = \sum_{\alpha=1}^{n} p_{\alpha} \dot{q}_{\alpha} - L$ [where p_{α} and $q_{\alpha} (\alpha = 1, 2, \dots, n)$ are the generalized momenta and coordinates respectively, i.e., $H = H(q_1, \dots, q_n, p_1, \dots, p_n, t)$, $L = L(q_1, \dots, q_n, \dot{q}_1, \dots, \dot{q}_n, t)$, $p_\alpha = \frac{\partial L}{\partial \dot{q}_\alpha}$ and $\dot{p}_\alpha = \frac{\partial L}{\partial q_\alpha}$] to show that (i) $\dot{q}_{\alpha} = \frac{\partial H}{\partial p_{\alpha}}$ $\alpha = 1, 2, \dots, n$ (4 marks) (ii) $\dot{p}_{\alpha} = -\frac{\partial H}{\partial q_{\alpha}}$ $\alpha = 1, 2, \dots, n$ (4 marks) (iii) $\frac{\partial H}{\partial t} = -\frac{\partial L}{\partial t}$ (7 marks) For a certain dynamical system the kinetic energy, T, and potential energy, V, are given by (b) $T = \dot{q}_1^2 + 2 \, \dot{q}_1 \, \dot{q}_2 + 3 \, \dot{q}_2^2$ $V = 4 q_1^2$ where q_1 , q_2 are the generalized coordinates. Find the momentum $p_1 \& p_2$ of the system. (2 marks) (i)

(ii) Use $H = \sum_{\alpha=1}^{L} p_{\alpha} \dot{q}_{\alpha} - L$ to find the Hamiltonian function of the system and

show that $H = \frac{1}{8} \left(3 p_1^2 - 2 p_1 p_2 + p_2^2 \right) + 4 q_1^2$ (8 marks)

. . .

Question two

85

(5 marks)

The definition of the Poisson brackets are given as $[u, v]_{q, p} \equiv \sum_{\alpha=1}^{n} \left(\frac{\partial u}{\partial q_{\alpha}} \frac{\partial v}{\partial p_{\alpha}} - \frac{\partial u}{\partial p_{\alpha}} \frac{\partial v}{\partial q_{\alpha}} \right)$, or simply written as [u, v], where q_{α} and p_{α} are the α^{th} generalized coordinate and

simply written as [u, v], where q_{α} and p_{α} are the α^{m} generalized coordinate and momentum respectively.

(a) For any function
$$F(q_1, q_2, \dots, q_n, p_1, p_2, \dots, p_n, t)$$
, prove that

$$\frac{dF}{dt} = [F, H] + \frac{\partial F}{\partial t}$$
where H is the Hamiltonian of the system, i.e., $H(q_1, q_2, \dots, q_n, p_1, p_2, \dots, p_n, t)$

(b) The three components of the angular momentum $\vec{l} (\equiv \vec{r} \times \vec{p})$ of a particle of mass m are given by $l_1 = q_2 p_3 - q_3 p_2$, $l_2 = q_3 p_1 - q_1 p_3$ and $l_3 = q_1 p_2 - q_2 p_1$ where $p_i = m \dot{q}_i$ i = 1, 2, 3. Show that

(i) $[l_1, l_2] = l_3$ (5 marks) (ii) $[q_2, l_3] = q_1$ (5 marks)

(c) For an equation of the type $\frac{du}{dt} = [u, H]$ the specific solution of u(t) is given by the following Taylor series expansion for the time t as $u(t) = u_0 + [u, H]_0 t + [[u, H], H]_0 \frac{t^2}{2!} + [[[u, H], H], H]_0 \frac{t^3}{3!} + \dots$ (1) where subscript 0 denotes the initial conditions at t = 0. For a simple harmonic oscillator system described by $H = \frac{p^2}{2m} + \frac{1}{2}kx^2$, if the initial conditions are given as x_0 and p_0 , use eq.(1) to deduce that

$$x(t) = x_0 + \frac{p_0}{m}t - \frac{k x_0}{2 m}t^2 - \frac{k p_0}{6 m^2}t^3 + \dots$$
 (10 marks)

Question three

$$L = T - V = \frac{1}{2} \mu \left(\dot{r}^{2} + r^{2} \dot{\theta}^{2} \right) + \frac{k}{r}$$

where μ is the reduced mass of the system, k is a positive constant and (r, θ) are polar coordinates of the motion plane with its origin at the center of mass of the two-body system.

(i) Write down the Lagrange's equation for θ and show that the angular momentum l is conserved, i.e., deduce that

$$\dot{\theta} = \frac{l}{\mu r^2}$$
 (1) where *l* is a constant. (3 marks)

(ii) Write down the Lagrange's equation for r, with eq.(1) inserted, deduce that $l^2 = k$ (2)

$$\mu \ddot{r} - \frac{1}{\mu r^3} + \frac{\pi}{r^2} = 0$$
(2) (3 marks)

(iii) Multiply eq.(2) by
$$dr$$
 and use $\ddot{r} dr = \frac{d\dot{r}}{dt} dr = d\dot{r} \frac{dr}{dt} = \dot{r} d\dot{r} = d\left(\frac{\dot{r}^2}{2}\right)$ to

show that the total energy $E (\equiv T + V)$ is conserved, i.e.,

$$\frac{1}{2}\mu\left(\dot{r}^2 + r^2\dot{\theta}^2\right) - \frac{k}{r} = const. \equiv E \quad \dots \qquad (3)$$
 (6 marks)

- (b) If an earth satellite of 500 kg mass is having a pure tangential speed $v_{\theta} = 8,000$ m/s at its near-earth-point 600 km above the earth surface,
 - (i) calculate the values of l and E of this satellite, (3 marks)
 - (ii) calculate the values of the eccentricity, ε , and show that the orbit is an elliptical orbit. Also calculate its period. (6 marks)
 - (iii) determine the value of the v_{θ} at the same given near-earth-point such that the satellite orbit is a circular orbit, (2 marks)

(Hint:
$$E = \frac{1}{2} \mu v_{\theta}^2 - \frac{k}{r} \xrightarrow{\text{circular orbit}} - \frac{k}{2r}$$
)

(iv) determine the value of the v_{θ} at the same given near-earth-point such that the satellite orbit is a parabolic orbit . (2 marks)

Question four

(a) Two set of Cartesian coordinate axes are having the same origins and z-axis. The nonprime system (referred to as "rotating" system) is rotating with an angular velocity $\vec{\omega} = \vec{e}_z \cdot \vec{\theta}$ about the prime system (referred as "fixed" system) as shown below:



For any vector field \vec{F} decomposed into the above two-set of cartesian components, i.e., $\vec{F} = \vec{e}_x F_x + \vec{e}_y F_y + \vec{e}_z F_z = \vec{e}_{x'} F_{x'} + \vec{e}_{y'} F_{y'} + \vec{e}_{z'} F_{z'}$, show that $\left(\frac{d\vec{F}}{dt}\right)_{fixed} = \left(\frac{d\vec{F}}{dt}\right)_{rotating} + \vec{\omega} \times \vec{F}$ where $\left(\frac{d\vec{F}}{dt}\right)_{fixed} = \vec{e}_{x'} \frac{dF_{x'}}{dt} + \vec{e}_{y'} \frac{dF_{y'}}{dt} + \vec{e}_{z'} \frac{dF_{z'}}{dt}$ and $\left(\frac{d\vec{F}}{dt}\right)_{rotating} = \vec{e}_x \frac{dF_x}{dt} + \vec{e}_y \frac{dF_y}{dt} + \vec{e}_z \frac{dF_z}{dt}$ (12 marks) (Hint : $\vec{e}_x = \vec{e}_{x'} \cos(\theta) + \vec{e}_{y'} \sin(\theta)$, $\vec{e}_y = -\vec{e}_{x'} \sin(\theta) + \vec{e}_{y'} \cos(\theta)$ and $\vec{e}_z = \vec{e}_{z'}$)



If a person, near the earth surface at a northern latitude λ , fired a bullet of speed, v_0 , at a target situated at his north direction $(-\vec{e}_x \text{ direction})$ of distance L away from him. Assuming he has a perfect rifle and the time T for the bullet hitting the target is short and $T \approx \frac{L}{v_0}$ (i.e., neglecting the gravitational bending and assuming the bullet is moving

along -x direction with constant speed v_0).

(i) Show that the bullet will miss the target by a deviation distance d resulting from the Coriolis force $(-2 \ m \ \vec{\omega} \times \vec{v}_r)$. Show that

$$d = \frac{\omega L^2}{v_0} \sin(\lambda)$$
(11 marks)
(Hint: $\vec{a}_{eff} \approx -2 \ \vec{\omega} \times \vec{v}_r$, $\vec{v}_r \approx \vec{e}_x (-v_0)$, $\vec{\omega} = \vec{e}_x (-\omega \cos(\lambda)) + \vec{e}_z (\omega \sin(\lambda))$)

(ii) Given the values of $\lambda = 60^{\circ}$, L = 2000 m, $v_0 = 800$ m/s and $\omega = 2\pi$ rad/day (i.e., $\omega = 7.27 \times 10^{-5}$ rad/s), determine the value of the deviation distance d. (2 marks)

Question five

Six equal mass point $m (= m_1 = m_2 = \dots = m_6)$ attached by massless rigid rods to form a rigid body of diamond shape with the center of mass of the "diamond" chosen as the origin of the body coordinate system (x_1, x_2, x_3) as shown in the diagram below.



where each mass point's coordinates in terms of length a & b is indicated in the diagram.

(a) Evaluate all elements of the inertia tensor, I, of the given rigid body with respect to the chosen body coordinate system and show that

$$I = \begin{pmatrix} 2m(a^{2}+b^{2}) & 0 & 0\\ 0 & 2m(a^{2}+b^{2}) & 0\\ 0 & 0 & 4ma^{2} \end{pmatrix}$$

(6 marks)

Question five (continued)

(b) If the given rigid body is only rotating with an angular velocity $\vec{\omega}$ without translational motion with respect to a fixed inertia coordinate system (x'_1, x'_2, x'_3) sharing the same origin as that of the body coordinate system, write down the total kinetic energy

$$T = T_{rotational} = \frac{1}{2} \vec{\omega} \bullet I \bullet \vec{\omega} \quad \text{in terms of} \quad m, a, b, \omega_1, \omega_2 \& \omega_3 \quad \text{where}$$
$$\vec{\omega} = \vec{e}_1 \omega_1 + \vec{e}_2 \omega_2 + \vec{e}_3 \omega_3 \quad . \tag{2 marks}$$

(c) The following are Euler's equations for force-free pure-rotational motion, i.e., $L = T_{rotational}$, for already diagonalized I as the case in (a).

$$\begin{cases} (I_2 - I_3) \,\omega_2 \,\omega_3 - I_1 \,\dot{\omega}_1 = 0 & \dots & (1) \\ (I_3 - I_1) \,\omega_3 \,\omega_1 - I_2 \,\dot{\omega}_2 = 0 & \dots & (2) \\ (I_1 - I_2) \,\omega_1 \,\omega_2 - I_3 \,\dot{\omega}_3 = 0 & \dots & (3) \end{cases}$$

(i) For our given rigid body, deduce from the above Euler's equations that

$$\omega_{3} = const. \xrightarrow{set as} K \qquad \dots \qquad (4)$$
$$\dot{\omega}_{1} = \frac{(-a^{2} + b^{2})K}{(a^{2} + b^{2})} \omega_{2} \qquad \dots \qquad (5)$$
$$\dot{\omega}_{2} = -\frac{(-a^{2} + b^{2})K}{(a^{2} + b^{2})} \omega_{1} \qquad \dots \qquad (6)$$

(6 marks)

(ii) If
$$b > a \& K > 0$$
, then $\frac{(-a^2 + b^2)K}{(a^2 + b^2)}$ is a positive constant and set
it as Ω . Deduce from eq.(5) and eq.(6) in (c)(i) that
 $\ddot{\omega}_1 = -\Omega^2 \omega_1 \quad \dots \quad (7)$ (3 marks)

- (iii) By direct substitution, show that $\omega_1 = A \cos(\Omega t + B)$ (8) is the solution to eq.(7) with A & B constant values linking to the given initial value of $\vec{\omega}$. (2 marks)
- (iv) Substitute eq.(8) into eq.(5) and deduce that $\omega_2 = -A \sin(\Omega t + B) \quad \dots \quad (9) \quad (2 \text{ marks })$
- (v) Show that the magnitude of $\vec{\omega}$ is a constant for all time t. (4 marks)

91

1

$$V = -\int \vec{F} \cdot d\vec{l} \quad and \ reversely \quad \vec{F} = -\vec{\nabla} V$$

$$L = T - V = L(q_1, q_2, \dots, q_n, \dot{q}_1, \dot{q}_2, \dots, \dot{q}_n, t)$$

$$p_a = \frac{\partial L}{\partial \dot{q}_a} \quad and \quad \dot{p}_a = \frac{\partial L}{\partial q_a}$$

$$H = \sum_{a=1}^{n} (p_a \dot{q}_a) - L = H(q_1, q_2, \dots, q_n, \dot{q}_1, \dot{q}_2, \dots, \dot{q}_n, t)$$

$$\dot{q}_a = \frac{\partial H}{\partial p_a} \quad and \quad \dot{p}_a = -\frac{\partial H}{\partial q_a}$$

$$[u, v] = \sum_{a=1}^{n} \left(\frac{\partial u}{\partial q_a} \frac{\partial v}{\partial p_a} - \frac{\partial u}{\partial p_a} \frac{\partial v}{\partial q_a}\right)$$

$$G = 6.673 \times 10^{-11} \quad \frac{N m^2}{kg^2}$$
radius of earth $r_E = 6.4 \times 10^6 m$
mass of earth $m_E = 6 \times 10^{24} kg$
earth attractive potential $= -\frac{k}{r}$ where $k = G m m_E$

$$\varepsilon = \sqrt{1 + \frac{2 E l^2}{\mu k}} \quad \{(\varepsilon = 0, \text{circle}), (0 < \varepsilon < 1, \text{ellipse}), (\varepsilon = 1, \text{parabola}), \dots\}$$

$$\mu = \frac{m_1 m_2}{m_1 + m_2} \approx m_1 \quad \text{if} \quad m_2 \gg m_1$$
For elliptical orbit, i.e., $0 < \varepsilon < 1$, then
$$\begin{cases} \text{semi-major } a = \frac{k}{2|E|} \\ \text{period } \tau = \frac{2\mu}{l} (\pi a b) \\ r_{\min} = a(1 - \varepsilon) \& r_{\max} = a(1 + \varepsilon) \end{cases}$$

for plane polar (r, θ) system with unit vectors $(\vec{e}_r, \vec{e}_{\theta})$, we have $\begin{cases} \vec{v} = \vec{e}_r \ \dot{r} + \vec{e}_{\theta} \ r \ \dot{\theta} \\ \vec{a} = \vec{e}_r \ (\vec{r} - r \ \dot{\theta}^2) + \vec{e}_{\theta} \ (2 \ \dot{r} \ \dot{\theta} + r \ \ddot{\theta}) \end{cases}$ $\vec{\nabla} f = \vec{e}_r \ \frac{\partial f}{\partial r} + \vec{e}_{\theta} \ \frac{1}{r} \ \frac{\partial f}{\partial \theta}$

• .

9

Useful informations (continued)

$$I = \begin{pmatrix} \sum_{\alpha} m_{\alpha} \left(x_{\alpha,2}^{2} + x_{\alpha,3}^{2} \right) & -\sum_{\alpha} m_{\alpha} x_{\alpha,1} x_{\alpha,2} & -\sum_{\alpha} m_{\alpha} x_{\alpha,1} x_{\alpha,3} \\ -\sum_{\alpha} m_{\alpha} x_{\alpha,2} x_{\alpha,1} & \sum_{\alpha} m_{\alpha} \left(x_{\alpha,1}^{2} + x_{\alpha,3}^{2} \right) & -\sum_{\alpha} m_{\alpha} x_{\alpha,2} x_{\alpha,3} \\ -\sum_{\alpha} m_{\alpha} x_{\alpha,3} x_{\alpha,1} & -\sum_{\alpha} m_{\alpha} x_{\alpha,3} x_{\alpha,2} & \sum_{\alpha} m_{\alpha} \left(x_{\alpha,1}^{2} + x_{\alpha,2}^{2} \right) \end{pmatrix}$$

 $\vec{F}_{eff} = \vec{F} - m \, \vec{\vec{R}}_f - m \, \vec{\vec{\omega}} \times \vec{r} - m \, \vec{\omega} \times (\vec{\omega} \times \vec{r}) - 2 \, m \, \vec{\omega} \times \vec{v}_r \qquad \text{where}$ $\vec{r}' = \vec{R} + \vec{r} \quad \text{and}$

- \vec{r}' refers to fixed (inertial system)
- \vec{r} refers to rotatinal (non inertial system) rotates with $\vec{\omega}$ to \vec{r} ' system
- \vec{R} from the origin of \vec{r} ' to the origin of \vec{r}
- $\vec{v}_r = \left(\frac{d\,\vec{r}}{d\,t}\right)_r$